A GENERAL METHOD TO COMPUTE NUMERICAL DISPERSION ERROR

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Abstract. This article presents a new methodology to compute numerical dispersion error. The analysis here presented is not restricted to uniform structured meshes nor linear discrete operators as it does not rely on sinusoids to compute the associated error. When using uniform meshes, the results obtained with the present method collapse onto the obtained with the classic one via an easy change of basis. If non-uniform meshes are used, a new kind of results are obtained which shed some light onto the role stretching has on dispersion error.

1 INTRODUCTION

Discretization of the different differential operators induces errors. Among others, the study of errors in the advective term in Computational Fluid Dynamics (CFD) or Computational AeroAcoustics (CAA) is still investigated. Diffusion error is totally avoided by preserving the skew-symmetry of the discrete operator, which is only achieved by using central schemes or Symmetry Preserving schemes analysed by Trias et al. [1], Verstapen and Veldman [2] or Rozema et al. [3].

Nevertheless, there are situations where keeping the skew-symmetry of the discrete operation, i.e. having null numerical diffusion, is not the prefered option. Sometimes numerical diffusion is introduced on purpose in order to avoid numerical instabilities. There are a wide number of options, from upwind numerical schemes to Essential Non-Oscillating, Weighted (WENO) [4] or not (ENO) [5], passing from TVD schemes and Flux or Slope Limiters [6].

However, even when the symmetry, or skew-symmetry, of the operator is preserved, there is another kind of unavoidable error: the dispersion error. Only if spectral methods are used, when the derivative is known and not approximated, the dispersion error can be avoided. Tam and Webb [7], Lele [8], Pirozzoli [9] and more recently Fauconnier and Dick [10] have studied the dispersion characteristics of several numerical schemes, ranging from linear ones to completely non-linear, such as previously mentioned WENO or Flux Limiter ones.
This kind of analysis has concluded that the use of high-order schemes is preferred over low-order ones. For example, in CAA the majority of its researchers use high-order schemes as the correct wave-propagation is the key to obtain acceptable results. In this sense, Bogey [11] and Shur et al. [12, 13] used high-order schemes in their jets simulations. Seo and Moon [14] and Ewert and Schröder [15] solved a different kind of wave equations based on Euler instead Navier-Stokes but had in common the usage of high-order schemes in their simulations. However, recent works in CAA have used low-order schemes to discretise the convective term of Navier-Stokes equations while obtaining acceptable results. Tyacke et al. [16] used the second order Kinetic Energy Preserving of Jameson [17] in the simulation of a jet. Fuchs simulated a single stream jet [18] and the Rudimentary Landing Gear (RLG) [19] using a second-order scheme and obtaining very good acoustic results in both simulations.

Another aspect which has to be taken into account, apart from the order of the numerical scheme, is the non-linearities that mesh introduces; this happens when non-uniform uniform meshes are used. None of the previous commented authors in CAA used uniform meshes in their simulations but stretched ones. Vichnevetsky [20] or Trefethen [21], among others, have studied how mesh irregularities, such as a non constant spacing, are the source of additional errors in simulations.

Thus, the use of high-order schemes in wave-propagation simulations is still not enough justified. Extrapolating the conclusions extracted in the conditions of the classical approach to compute dispersion, i.e. uniform structured meshes and linear schemes, to a general framework could not be the best recommended course of action. In this work we develop a new methodology that avoids the limitations of the previous analysis and allows to evaluate any kind of mesh with any kind of numerical scheme.

2 METHODOLOGY

2.1 Analytical background

The Fourier Transform of a function \( f(x) : \mathbb{R} \rightarrow \mathbb{R} \) is given by:

\[
\hat{f} = \int_{-\infty}^{\infty} f(x) e^{-2\pi i x \alpha} dx.
\]  

(1)

The Fourier Transform of the spatial derivative of \( f(x) \) is:

\[
\hat{f}'(\alpha) = i \alpha \hat{f}.
\]  

(2)

Reordering the last equation, is easy to isolate the analytical wavenumber as a function of \( f \) and its derivative:

\[
\alpha = -i \frac{\hat{f}'(\alpha)}{f}.
\]  

(3)
Nevertheless, when the space is discretised, previous equation does not hold. Applying both derivative and shifting theorems as Tam [7, 22], the discrete equivalent of Eq.(3) is obtained:

\[ \tilde{\alpha} = \frac{-i}{\Delta x} \left[ \sum_k a_k e^{ik\alpha \Delta x} \right]. \] (4)

As can be seen, previous definition implies a uniform spacing as a single \( \Delta x \) appears. Thus, as the method requires using Fourier Transform properties, this analysis is restricted only to uniform meshes. However, we can see the Fourier Transform as a projection. Using the idea to project the function onto a set of orthonormal functions, we can avoid the limitations that the classical method has.

First, let’s define the inner product of \( \{ \xi, \psi \} \in L^2(\Omega_\mu, \mu) \), where \( L^2(\Omega_\mu, \mu) \) is the space of square Lebesgue-integrable functions \( \psi : \Omega_\mu \subset \mathbb{R} \mapsto \mathbb{C} \) as:

\[ \langle \xi | \psi \rangle_{\Omega_\mu} = \int_{\Omega_\mu} \xi(\mu) \overline{\psi(\mu)} d\mu. \] (5)

Then, let \( \Phi = \{ \phi_{-N}(x), \phi_{-N+1}(x), \ldots, \phi_{-1}(x), \phi_0(x), \phi_1(x), \phi_2(x), \ldots, \phi_N(x) \} \) be an orthonormal basis of functions of \( \Omega_\times \subset \mathbb{R} \), i.e,

\[ \langle \phi_m | \phi_n \rangle_{\Omega_\times} = \delta_{mn}, \] (6)

where \( \delta_{mn} \) is the Kronecker’s delta.

We can define a mapping \( T : L^2(\Omega_\times, x) \mapsto \mathbb{C}^{2N+1}; T : f(x) \mapsto (\alpha_m) \in \mathbb{C}^{2N+1} \), where

\[ \alpha_m = \langle f | \phi_m \rangle_{\Omega_\times} = \int_{\Omega_\times} f(x) \overline{\phi_m(x)} dx, \] (7)

is the projection of \( f(x) \) onto the \( m \) function \( \Phi \). Under the pertinent smoothness of \( f \) criterion,

\[ f(x) \simeq S_N = \sum_{m=-N}^N \alpha_m \phi_m(x); \quad \lim_{N \to \infty} S_N = f(x). \] (8)

This defines the inverse mapping \( T^{-1} : \mathbb{C}^{2N+1} \mapsto L^2(\Omega_\times, x), T : (\alpha_m) \in \mathbb{C}^{2N+1} \mapsto f(x). \)

The derivative \( f' = \frac{d}{dx} f(x) \) can also be expressed in terms of its projection on the set of functions of \( \Phi \):

\[ f'(x) \simeq S'_N = \sum_{m=-N}^N \alpha_m \phi'_m(x) \simeq \sum_{m=-N}^N \left( \alpha_m \sum_{n=-N}^N \gamma_{mn} \phi_n(x) \right), \] (9)

where

\[ \gamma_{mn} = \langle \phi'_m | \phi_n \rangle_{\Omega_\times}. \] (10)
We define the matrix $\Gamma$, where its elements $(\Gamma)_{mn} = \gamma_{mn}$ represent the projections of the derivatives of $\phi_m$ with respect to “$x$” on $\phi_n$.

$\Gamma$ characteristics are determined by the selected basis of functions $\Phi$ and the properties of the derivative operator. Assuming null contributions from boundaries and integrating by parts, it is straightforward to show that the derivative operator is skew-Hermitian with respect to the inner product of Eq.(5), i.e.,

$$\left\langle \frac{d}{dx} \xi | \psi \right\rangle_{\Omega_x} = -\left\langle \xi | \frac{d}{dx} \psi \right\rangle_{\Omega_x} . \quad (11)$$

Thus, $\Gamma$ is skew-Hermitian too.

Using an approximate derivative operator $\tilde{\frac{d}{dx}}(\cdot)$, one gets $\tilde{\Gamma} \in \mathbb{C}^{2N+1 \times 2N+1}$ instead of $\Gamma$. The errors associated with using approximate differential operators to approximate derivatives are the deviations of $\tilde{\gamma}_{mn}$ from $\gamma_{mn}$. Finally, to ease the analysis, $\Phi$ can be chosen such that $\Gamma$ is diagonal and known, i.e., $\phi_m$ is an eigenfunction of the derivative. Then, the possible errors in $\tilde{\Gamma}$ are:

- $\tilde{\gamma}_{mn} \neq 0$ if $m \neq n$,
- $\text{Re}(\tilde{\gamma}_{mm}) \neq 0$, and
- $\text{Im}(\tilde{\gamma}_{mm}) \neq 1$,

where $\lambda_m \in \mathbb{II}$ is the eigenvalue of the derivative on $\phi_m$.

### 2.2 Numerical method

Defining the discrete differential operator as $A$, and splitting it into the Hermitian, $D$, and skew-Hermitian, $C$, parts [23],

$$C = \frac{1}{2} (A - A^*); \quad (12)$$

$$D = \frac{1}{2} (A + A^*); \quad (13)$$

$$A = C + D, \quad (14)$$

where $(\cdot)^*$ indicates the conjugate transpose. The matrices $C$ and $D$ that appear after splitting the discrete differential operator have interesting properties involving the inner product:

$$\left\langle C \psi | \psi \right\rangle \in \mathbb{II}$$

$$\left\langle D \psi | \psi \right\rangle \in \mathbb{R}$$

$$\left\langle C \psi | \eta \right\rangle = -\left\langle \psi | C \eta \right\rangle$$

$$\left\langle D \psi | \eta \right\rangle = \left\langle \psi | D \eta \right\rangle , \quad (15)$$

where $\{\psi, \eta\} \in \mathbb{C}^N$. 


This matrices and Γ are closely related. More precisely, matrix C is related to the real part of the scalars γ that have been defined on Eq.(10). D is related to the imaginary part of these values. When a discretization is used, i.e. a matrix A is defined, these scalars can be numerically calculated:

$$\tilde{\gamma}_{mn} = \langle A\phi_m | \phi_n \rangle$$
$$\tilde{c}_{mn} = \text{Im}(\tilde{\gamma}_{mn}) = \langle C\phi_m | \phi_n \rangle$$
$$\tilde{d}_{mn} = \text{Re}(\tilde{\gamma}_{mn}) = \langle D\phi_m | \phi_n \rangle.$$  (16)

This development holds because Φ is orthonormal and due to the nature of the skew and Hermitian operators. They can be calculated from A:

$$\tilde{\gamma}_{mn} = \langle C\phi_m | \phi_n \rangle = \langle A\phi_m | \phi_n \rangle - \langle \phi_n | A\phi_m \rangle$$
$$\tilde{d}_{mn} = \langle D\phi_m | \phi_n \rangle = \langle A\phi_m | \phi_n \rangle + \langle \phi_n | A\phi_m \rangle,$$  (17)

where the values of $$\tilde{\gamma}_{mn}$$ are the different elements of the matrix $$\tilde{\Gamma}.$$ To simplify the analysis, the root mean square of the second index of the values of $$\tilde{\gamma}$$ can be calculated,

$$\tilde{\lambda}_C^m = \sqrt{\sum_n \left(\frac{\langle A\phi_m | \phi_n \rangle - \langle \phi_n | A\phi_m \rangle}{2}\right)^2}$$
$$\tilde{\lambda}_D^m = \sqrt{\sum_n \left(\frac{\langle A\phi_m | \phi_n \rangle + \langle \phi_n | A\phi_m \rangle}{2}\right)^2}.$$  (18)

This procedure allows a faster comparison between the analytical value and the recovered numerical ones. Thus, if the recovered $$\tilde{\lambda}_D^m$$ is not null, the differential operator will have diffusive behavior. If the ratio between $$|\tilde{\gamma}_{mn}|$$ and $$\tilde{\lambda}_C^m$$ is smaller than the unity, then the off-diagonal values of $$\tilde{\Gamma}_{mn},$$ which should be zero, will have a non-zero value. And finally, the ratio between $$\tilde{\lambda}_m$$ and the reference parameter indicates the deviation of the numerical discretization from its expected value.

2.2.1 Orthonormal basis

Once the method is defined, the next step is choosing an appropriate orthonormal basis. Here we propose to use the eigenvectors of the discrete Laplacian operator. The main reasons can be summarized as:

- They retain the concept of mesh connectivity as they are calculated from the discrete Laplacian, which takes into account how mesh nodes are linked between them.
- In evenly spaced domains, the eigenvectors of the discrete Laplacian are, indeed, discrete sinusoids.
• And finally, in the continuous limit, or as we refine the mesh, the eigenvectors and eigenvalues of the discrete Laplacian collapse onto the eigenfunctions and eigenvectors of the continuous Laplacian, i.e. sinusoids.

Additionally, the discrete Laplacian $L$ and the discrete gradient $G$ are related [24]:

$$L = G^* G.$$  \hspace{1cm} (19)

The Singular Value Decomposition of $G$ relates its right eigenvectors and singular values with the right eigenvectors and eigenvalues of $L$. More precisely, $G$ and $L$ share the same eigenvectors and the eigenvalues of $G$ are the square-root of the eigenvalues of $L$.

### 2.2.2 Phase effect

When non-linear schemes or non-uniform meshes are used, the derivative is distorted. More precisely, the derivation process projects each eigenmode onto several components instead a single one. In uniform meshes and using non-linear discrete schemes, Fauconnier and Dick [10] tackled this problem by setting the phase inside the sinusoids in the Fourier Transform as a random parameter; their idea was that the numerical wavenumber is a function of both analytical wavenumber and the random phase. Repeating the process 5000 times, they finally computed an averaged numerical wavenumber which did not depend on the phase.

In our case this is equivalent to a rotation with random angle of each pair of eigenvectors that conforms the base, with the rotation matrix being:

$$R = \begin{pmatrix}
1 & 0 & 0 & \cdots & 0 & 0 & 0 \\
0 & \cos(\alpha_1) & \sin(\alpha_1) & \cdots & 0 & 0 & 0 \\
0 & -\sin(\alpha_1) & \cos(\alpha_1) & \cdots & 0 & 0 & 0 \\
0 & 0 & 0 & \cdots & 0 & 0 & 0 \\
0 & 0 & 0 & \cdots & 0 & 0 & 0 \\
& \ddots & \ddots & \ddots & \ddots & \ddots & \ddots \\
0 & 0 & \cdots & \cos(\alpha_{N-1}) & \sin(\alpha_{N-1}) & 0 & 0 \\
0 & 0 & \cdots & -\sin(\alpha_{N-1}) & \cos(\alpha_{N-1}) & 0 & 0 \\
0 & 0 & \cdots & 0 & 0 & \cos(\alpha_N) & \sin(\alpha_N) \\
0 & 0 & \cdots & 0 & 0 & -\sin(\alpha_N) & \cos(\alpha_N)
\end{pmatrix}.$$ \hspace{1cm} (20)

The eigenvectors in the rotated base will be:

$$E' = RE,$$ \hspace{1cm} (21)

where $E'$ are the eigenvectors in the rotated basis and $E$ the eigenvectors in the original one.

Differently than with uniform meshes, where each Laplacian eigenvalue appears twice (except 0 and the maximum one for meshes with even number of nodes), non-uniform meshes does not have this property. This means that an effective eigenvalue is computed: the average between two consecutive eigenvalues that are associated with the two rotated eigenvectors.
Even though the procedure developed here could be used in any kind of mesh, such as 3D Cartesian or unstructured, it is important to take into account two critical facts:

- First, structured meshes are the most common approach in the CAA community. So, it would make sense to study the same kind of meshes that are used by the majority. Additionally, working with fully unstructured meshes implies that the recovered eigenvalue cannot be decomposed onto its Cartesian components. In other words, we will recover the wavenumber at specific directivity angle, but will have no way of knowing this angle.

- Second, if structured meshes are used, the effective wavenumber in multiple dimensions is, in fact, the combination of the different wavenumbers in the set of dimensions that conform the problem. So, it makes sense to simplify the problem and just to study a 1D problem.

Consequently, the meshes used in this article are just structured one-dimensional meshes. However, stretching can be studied with the methodology that we have presented. Therefore, the mesh set will include both uniform and non-uniform meshes. The mesh characteristics can be summarized as:

- 6 different Poisson ratios: from 0 to 5%.
- 5 different minimum mesh sizes: from 1/32 to 1/512.

Regarding the convective schemes, several of both linear and non-linear ones will be studied. For the sake of clarity, in Table 1 we include the acronyms of the different convective schemes that have been studied. However, we do not include the legend of meshes; the reasons will be commented with the results.

3.1 Dispersion error results

In Figures 1 and 2 we have included the recovered eigenvalue, using Eq.(18) and the concept of basis rotation, for several schemes; then, this value is multiplied by the maximum mesh size.

<table>
<thead>
<tr>
<th>Acronym</th>
<th>Convective scheme</th>
<th>Acronym</th>
<th>Convective scheme</th>
</tr>
</thead>
<tbody>
<tr>
<td>UPW</td>
<td>First-order upwind</td>
<td>MM</td>
<td>Minmod flux-limiter</td>
</tr>
<tr>
<td>SP2</td>
<td>Second-order symmetric</td>
<td>SB</td>
<td>Superbee flux-limiter</td>
</tr>
<tr>
<td>DRP4</td>
<td>Fourth-order DRP</td>
<td>VL</td>
<td>Van Leer flux-limiter</td>
</tr>
<tr>
<td>DRP6</td>
<td>Sixth-order DRP</td>
<td>WENO3</td>
<td>Third-order WENO</td>
</tr>
<tr>
<td>SP6</td>
<td>Sixth-order symmetric</td>
<td>WENO5</td>
<td>Fifth-order WENO</td>
</tr>
<tr>
<td>MLS3</td>
<td>Sixth-order Moving Least-Squares</td>
<td>WENO7</td>
<td>Seventh-order WENO</td>
</tr>
</tbody>
</table>

Table 1: Acronyms of the convective schemes.
As can be seen in these figures, only two kind of plots are observed: in solid line, results corresponding to uniform meshing; the dots, belong to stretched meshes. Data show that stretched meshes collapse onto the same plot, independently of the stretching factor used. There is, indeed, a fast transition from the uniform plot towards the stretched one.

In Tables 2 and 3 we have included the evolution of the peak in these plots, which is a signal from the transition from the uniform to the stretched plot. Low-order schemes exhibit a slow transition from uniform plot towards the stretched one, as the maximum peak is achieved for highly stretched meshes. On the other hand, high-order schemes almost do not have this transition; for slightly stretched meshes they achieve the maximum peak and the kind of plot typical for stretched meshes. Therefore, it seems that high-order schemes are more sensitive to mesh transformations, as they behave more differently faster than low-order schemes.
And finally, all stretched plots share the same common feature: around $\lambda \Delta_{\text{Max}} \sim 2$ there is a sudden drop of the recovered value; this is an indicative of the maximum allowed mode in the mesh, which just depends on the maximum mesh size. And this happens just with the eigenvalues, not the frequencies. In Figure 3 we include the recovered non-dimensional wavenumber for all the studied meshes and different schemes. In this case, there are not just two single plots, but a lot of them. Consequently, there is not a “maximum frequency”, which is a function of the maximum mesh size, as the previously commented drop does not appear on these new plots.

<table>
<thead>
<tr>
<th>Stretch. [%]</th>
<th>WENO3</th>
<th>WENO5</th>
<th>WENO7</th>
<th>MM</th>
<th>SB</th>
<th>VL</th>
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</thead>
<tbody>
<tr>
<td>0</td>
<td>1.1667</td>
<td>1.3317</td>
<td>1.4529</td>
<td>1.2526</td>
<td>1.4053</td>
<td>1.3237</td>
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<td>1</td>
<td>1.3747</td>
<td>1.52</td>
<td>1.6315</td>
<td>1.4713</td>
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<tr>
<td>2</td>
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<td>1.5402</td>
<td>1.6432</td>
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<td>1.6444</td>
<td>1.566</td>
</tr>
<tr>
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</tr>
<tr>
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<tr>
<td>5</td>
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<td>1.5374</td>
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<tr>
<td>AVG</td>
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<td>1.5289</td>
<td>1.6301</td>
<td>1.4867</td>
<td>1.6327</td>
<td>1.553</td>
</tr>
</tbody>
</table>

Table 3: Non-dimensional maximum eigenvalue normalized respect to the maximum eigenvalue for first-order upwind in uniform meshes, non-linear schemes.
4 CONCLUSIONS

In this article, a new methodology to compute numerical dispersion error not restricted to uniform meshes has been presented. Instead of requiring the use of Fourier Transform to evaluate the numerical wavenumber, we use the eigenvectors of the discrete Laplacian as the orthonormal base.

Non-dimensional results of the recovered eigenvalue via the maximum mesh size show that two different plots are recovered: one for uniform meshing and other for stretched meshes, independently of the studied stretching factor the plots obtained collapse onto a single one.

In the studied plots for stretched meshes, a sudden drop in the recovered eigenvalue, i.e. an increase of the numerical error, always appears. This drop is always located around $\lambda \Delta_{\text{Max}} \sim 2$. However, recovered non-dimensional wavenumber, via base change, do not exhibit this drop. Moreover, the plots for different meshes are a function of both maximum mesh size and used stretching factor, as the results do not collapse onto a single plot. This reinforces the idea that stretched meshes have a maximum allowed eigenvalue rather than a maximum allowed frequency.

As a further work, we would like that the herepresented methodology could help to select the most appropriate scheme for a given mesh geometry or be used to design meshing techniques leading to a reduction in the dispersion error.
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