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Turbulent flow around a wall-mounted cube: DNS and regularization modelling

<u>F.X. Trias</u>^{*,*}, A. Gorobets^{*}, R.W.C.P. Verstappen^{*}, M. Soria^{*} and A. Oliva^{*}

*Centre Tecnològic de Transferència de Calor (CTTC), Technical University of Catalonia C/ Colom 11, 08222 Terrassa, Barcelona, Spain, E-mail: cttc@cttc.upc.edu

*Institute of Mathematics and Computing Science, University of Groningen P.O. Box 800, 9700 AV Groningen, The Netherlands, E-mail: R.W.C.P.Verstappen@rug.nl





Presentation outline

1. Introduction

- Problem definition: Wall-mounted cube in a channel flow
- Parallel fully-3D Poisson solver
- DNS results for $Re_h = 7235$
- Governing equations

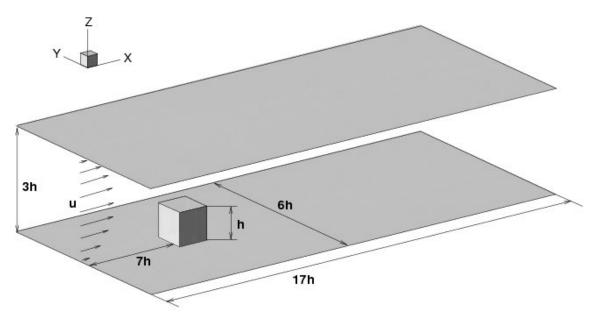
2. Regularization models for the simulation of turbulence

- Existing regularization: Leray and Navier-Stokes- α models
- Symmetry-preserving regularization models
- Mathematical foundation
- Discretizing the C_n regularization modelling
- Previous experience with C_4 regularization modelling
- 3. Results for a Turbulent flow around a Wall-mounted Cube
 - Description of cases
 - Comparison with DNS results
- 4. Conclusions and Future Research





Problem definition: Wall-mounted cube in a channel flow



Boundary conditions:

- Spanwise: Periodic BC
- Streamwise: prescribed inlet profile
- Non-slip BC on the channel walls and one the surface of the cube (No IB method is used!!)

Dimensionless governing numbers:

•
$$Re_h = U_{bulk}h/\nu = 7235 \dots$$

• ...or $Re_{\tau} = 590$

$$U/u_{\tau} = min(y^+, klny^+ + B)$$

where $y^+ = (y/H)Re_{\tau}$, $u_{\tau} = Re_{\tau}\nu/H$ and H = 3h/2, k = 0.25, B = 5.0.





Parallel fully-3D Poisson solver

A two-level Multigrid method

Algorithm on the i-th iteration:

- 1. Smoother: $\mathbf{x}_i^{3D} \approx \left(\mathbf{A}^{3D}\right)^{-1} \mathbf{b}^{3D}$
 - Locally preconditioned CG.
 - Initial guess \mathbf{x}_{i-1}^{3D}

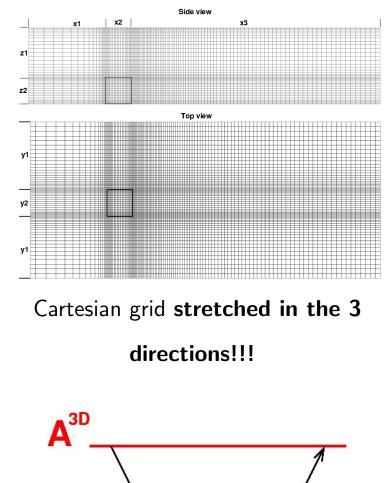
2.
$$\mathbf{r}_i^{3D} = \mathbf{A}^{3D} \mathbf{x}_i^{3D} - \mathbf{b}^{3D}$$

3. $\mathbf{r}_i^{2.5D} = \mathbf{Q} \mathbf{r}_i^{3D}$

- 4. Error equation: $\mathbf{z}_i^{2.5D} \approx (\mathbf{A}^{2.5D})^{-1} \mathbf{r}_i^{2.5D}$ A 2.5D parallel Poisson solver is used
 - FFT decompose $\mathbf{A}^{2.5D}$ into a set of 2D systems: $\hat{\mathbf{A}}_{k}^{2D}\hat{\mathbf{z}}_{k}^{2D} = \hat{\mathbf{r}}_{k}^{2D}$
 - $\hat{\mathbf{z}}_{k}^{2D} \approx \left(\hat{\mathbf{A}}_{k}^{2D}\right)^{-1} \hat{\mathbf{r}}_{k}^{2D}$ with $k = 1, \cdots, N_{x}$ A 2D parallel solver is used.
 - Inverse FFT of $\hat{\mathbf{z}}_k^{2D}$ gives $\mathbf{z}_i^{2.5D}$.

5.
$$\mathbf{z}_{i}^{3D} = \mathbf{P}\mathbf{z}_{i}^{2.5D}$$

6. $\mathbf{x}_{i+1}^{3D} = \mathbf{x}_{i}^{3D} + \mathbf{z}_{i}^{3D}$



2.5D

D

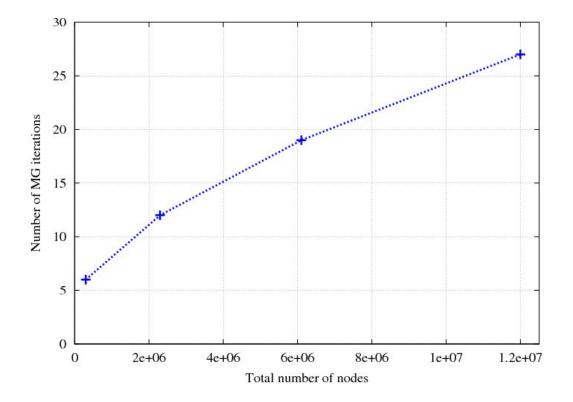




Parallel fully-3D Poisson solver

Parallel performance

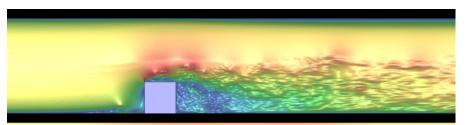
- $Re_h = 5000$
- PCG: 15 iterations using Jacobi as preconditioner
- Most of the time is consumed solving $\mathbf{A}^{2.5D}\mathbf{z}_i^{2.5D} = \mathbf{r}_i^{2.5D}$ therefore number of MG iterations is a good measure of how many times slower is the 3D solver respect the 2.5D

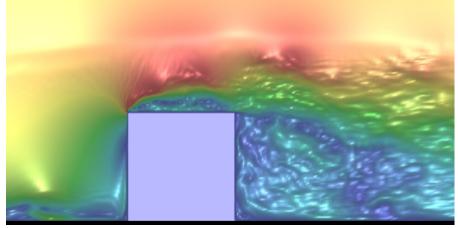






DNS results at $Re_h = 7235$





Some details about **DNS simulation**:

- Fully-3D Poisson solver
- Mesh size: $400 \times 200 \times 196 \approx 16M$
- Computing Time: $\approx 1 \text{ month} 300 \text{ CPUs}$ on MareNostrum supercomputer
- Symmetry-preserving discretization

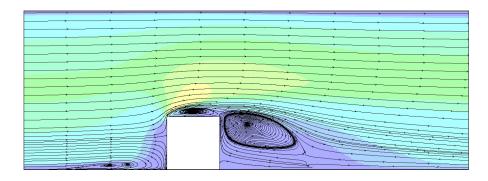
Complexity of the flow:

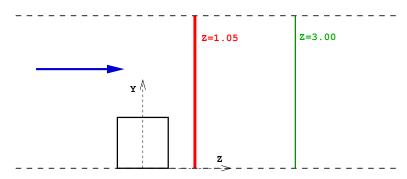
- Vortical structures:
 - ⋆ horseshoe-type at the upstream face
 - \star arc-shaped in the wake
- Flow separation
- Vortex shedding

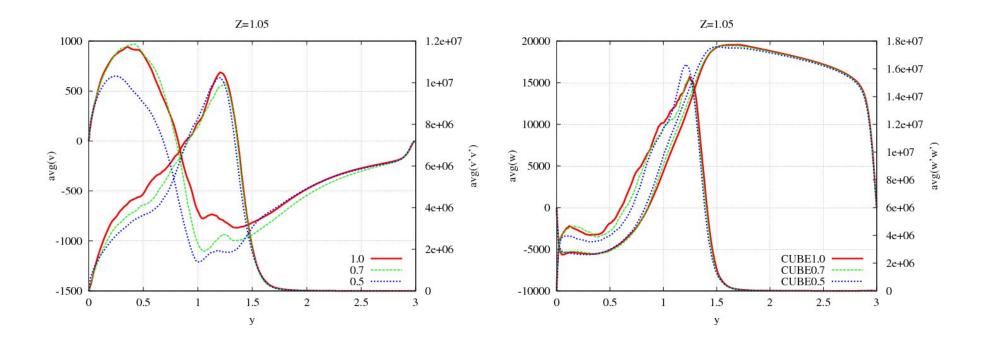




Verification of simulation



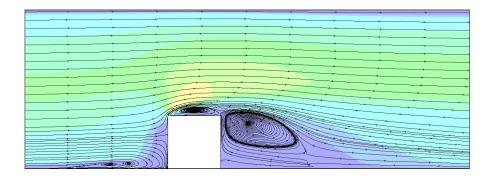


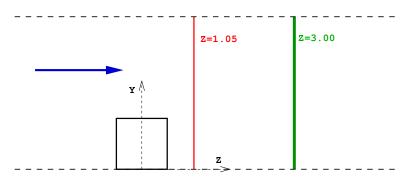


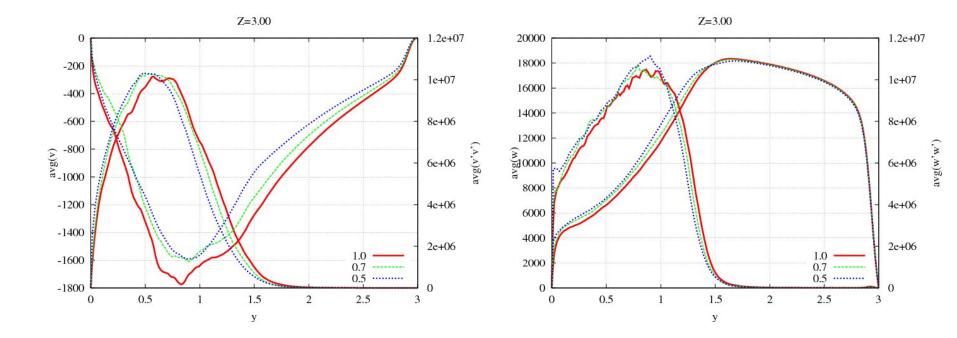




Verification of simulation



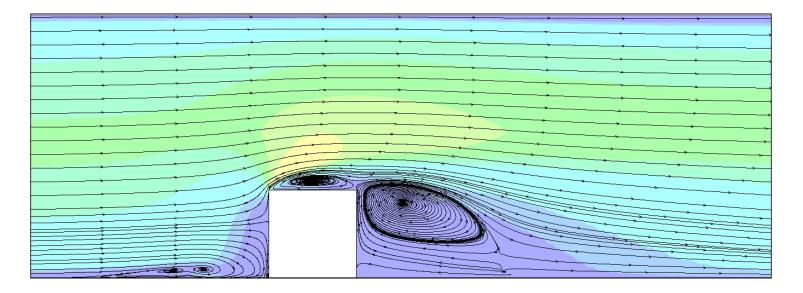


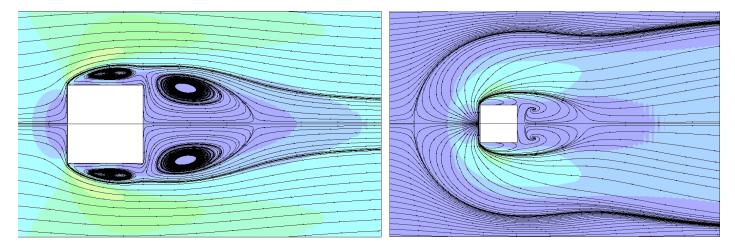






Streamlines









Governing equations

Incompressible Navier-Stokes equations

$$\partial_t u + \mathcal{C}(u, u) = Pr\mathcal{D}(u) - \nabla p + f$$

 $\nabla \cdot u = 0$

where the **nonlinear convective term** is given by

$$\mathcal{C}(u,v) = (u \cdot \nabla)v$$

and the linear dissipative term is given by

$$\mathcal{D}(u) = rac{1}{Re}
abla^2 u$$





Regularization modelling

As the full energy spectrum cannot be computed, a **dynamically less complex mathematical formulation** is sought. We consider smooth approximations (regularizations) of the nonlinearity,

$$\partial_t u_{\epsilon} + \widetilde{\mathcal{C}}(u_{\epsilon}, u_{\epsilon}) = \mathcal{D}(u_{\epsilon}) - \nabla p_{\epsilon} + f$$

such approximations may fall in the Large-Eddy Simulation (LES) concept,

$$\partial_t \bar{u}_\epsilon + \mathcal{C}(\bar{u}_\epsilon, \bar{u}_\epsilon) = \mathcal{D}(\bar{u}_\epsilon) - \nabla \bar{p}_\epsilon + f + \mathcal{M}_1(\bar{u}_\epsilon, \bar{u}_\epsilon)$$

if the filter is invertible:

$$\mathcal{M}_1(ar{u}_\epsilon,ar{u}_\epsilon) \ \ = \ \ \mathcal{C}(ar{u}_\epsilon,ar{u}_\epsilon) - \overline{\widetilde{C}(u_\epsilon,u_\epsilon)}$$





Previous regularization modellings

Leray and Navier-Stokes- α models

The regularization methods basically **alters the convective term** to **restrain the production of small scales** of motion.

• Leray model:

$$\partial_t u_\epsilon + \mathcal{C}(\bar{u}_\epsilon, u_\epsilon) = \mathcal{D}(u_\epsilon) - \nabla p_\epsilon$$

• Navier-Stokes- α model:

$$\partial_t u_\epsilon + \mathcal{C}_r(u_\epsilon, \bar{u}_\epsilon) = \mathcal{D}(u_\epsilon) - \nabla \pi_\epsilon$$

where the $\pi = p + u^2/2$ and the convetive operator in rotational form is defined as

$$\mathcal{C}_r(u,v) = (\nabla \times u) \times v$$

However, in doing so some of the **inviscid invariants** (kinetic energy, enstrophy in 2D and helicity in 3D) are **not conserved**.





Symmetry-preserving regularization models (1/2)

In order to conserve the following inviscid invariants

• Kinetic energy

• Helicity (in 3D)

Enstrophy (in 2D)

 $\int_{\Omega} (
abla imes oldsymbol{u}) \cdot oldsymbol{u} d\Omega$

the approximate convective operator has to be skew-symmetric:

$$\left(\widetilde{\mathcal{C}}(u,v),w\right) = -\left(\widetilde{\mathcal{C}}(u,w),v\right)$$





Symmetry-preserving regularization models (2/2)

This criterion yields the following class of approximations,

$$\partial_t u_\epsilon + \mathcal{C}_n(u_\epsilon, u_\epsilon) = \mathcal{D}(u_\epsilon) - \nabla p_\epsilon$$

in which the convective term in smoothened according to:

$$\begin{aligned} \mathcal{C}_2(u,v) &= \overline{\mathcal{C}(\bar{u},\bar{v})} \\ \mathcal{C}_4(u,v) &= \mathcal{C}(\bar{u},\bar{v}) + \overline{\mathcal{C}(\bar{u},v')} + \overline{\mathcal{C}(u',\bar{v})} \\ \mathcal{C}_6(u,v) &= \mathcal{C}(\bar{u},\bar{v}) + \mathcal{C}(\bar{u},v') + \mathcal{C}(u',\bar{v}) + \overline{\mathcal{C}(u',v')} \end{aligned}$$

where $u' = u - \bar{u}$ and $C_n(u, v) = C(u, v) + O(\epsilon^n)$ for any symmetric filter.





Mathematical foundation

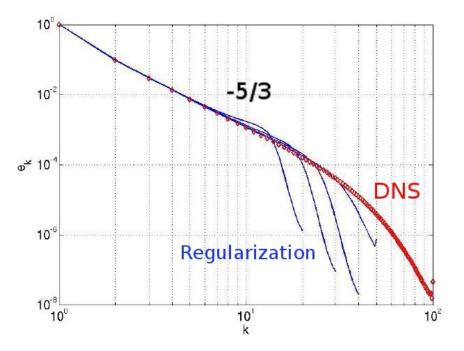
Energy flux equation for the symmetry-preserving regularization resembles the NS

$$\frac{1}{2}\frac{d}{dt}\left|u_{kk'}\right|^{2}+\nu\left|\nabla u_{kk'}\right|^{2}=\widetilde{T}_{k}-\widetilde{T}_{k'}\quad\longrightarrow\quad\nu<\left|\nabla u_{kk'}\right|^{2}>=<\widetilde{T}_{k}>-<\widetilde{T}_{k'}>$$

 \implies Following the same steps as Foias *et al.* (2001)

- $< \widetilde{T}_k >$ is a nonnegative, monotone decreasing function.
- $< \widetilde{T}_k >$ is approximately constant for $k_a < k < k_b$ (existence of inertial range).

 $\implies -5/3$ scaling !!!







LES-interpretation of C_4 -regularization

$$\partial_t \bar{u}_{\epsilon} + \mathcal{C}(\bar{u}_{\epsilon}, \bar{u}_{\epsilon}) - \mathcal{D}(\bar{u}_{\epsilon}) + \nabla \bar{p}_{\epsilon} =$$

$${\cal C}(ar u_\epsilon,ar u_\epsilon)-\overline{{\cal C}_4(u_\epsilon,u_\epsilon)} ~=~$$

$$-\frac{\epsilon^2}{12} \nabla \cdot (\nabla \bar{u}_{\epsilon} : \nabla \bar{u}_{\epsilon}) + \mathcal{O}(\epsilon^4)$$

gradient model + stabilization



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Discretizing the C_n regularization modelling

• The discretization is also a regularization. The **spatial discretization** method preserves the symmetry and conservation properties too

$$\Omega_{s}\frac{d\boldsymbol{u}_{s}}{dt} + \mathsf{C}\left(\boldsymbol{u}_{s}\right)\boldsymbol{u}_{s} + \mathsf{D}\boldsymbol{u}_{s} + \Omega_{s}\mathsf{G}\boldsymbol{p}_{c} = \boldsymbol{0}_{s} \qquad \text{with} \quad \mathsf{C}\left(\boldsymbol{u}_{s}\right) = -\mathsf{C}^{*}\left(\boldsymbol{u}_{s}\right)$$

and is therefore well-suited to test the proposed regularization model.

• A normalized self-adjoint filter has been chosen. In 1D it becomes

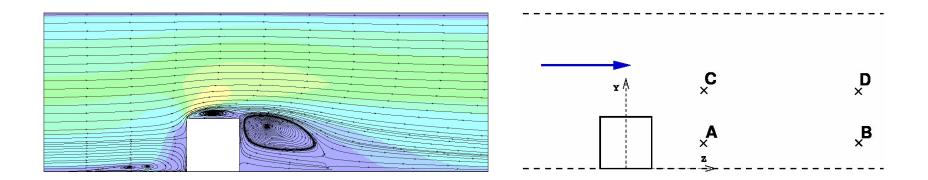
$$\overline{\phi}_{i} = \frac{\epsilon^{4} - 4\epsilon^{2}}{1152} \left(\phi_{i+2} + \phi_{i-2}\right) + \frac{16\epsilon^{2} - \epsilon^{4}}{288} \left(\phi_{i+1} + \phi_{i-1}\right) + \frac{\epsilon^{4} - 20\epsilon^{2} + 192}{192} \phi_{i}$$

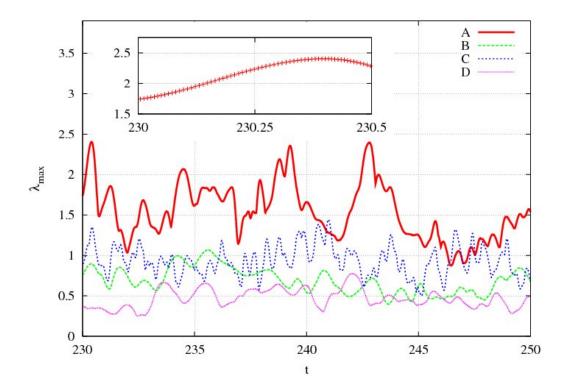
Thus, only one parameter needs to be prescribed: the local filter length ϵ !!!





Parameter-free approach









Parameter-free approach

The vortex-stretching and dissipation term contributions to $(1/|\omega|^2)\partial_t |\omega|^2$ are given by

$$\frac{\omega \cdot \mathcal{C}(\omega, u)}{\omega \cdot \omega} = \frac{\omega \cdot \mathcal{S}(u) \, \omega}{\omega \cdot \omega} \quad \text{and} \quad \frac{1}{Re} \frac{\nabla \omega : \nabla \omega}{\omega \cdot \omega}$$

At the smallest grid scale, $k = \pi/h$, convection may dominate diffusion

$$\frac{\omega_{k} \cdot \mathcal{C} (\omega, u)_{k}}{\omega_{k} \cdot \omega_{k}} > \frac{1}{Re} k^{2}$$

 \implies In the present work we **determine the filter width** ϵ from

$$\frac{\omega_k \cdot \mathcal{C}_4 \left(\omega, u\right)_k}{\omega_k \cdot \omega_k} \quad \approx \quad \frac{1}{Re} k^2$$





Parameter-free approach

Note that $C_4(u, v)$ depends on the filter length ϵ . For the smallest scale this dependence becomes

$$\frac{\omega_k \cdot \mathcal{C}_4 (\omega, u)_k}{\omega_k \cdot \omega_k} \approx f_4 (\hat{g}_k(\epsilon)) \frac{\omega_k \cdot \mathcal{S} (u) \omega_k}{\omega_k \cdot \omega_k} \leq f_4 (\hat{g}_k(\epsilon)) \frac{\lambda_{max}}{(\omega_k \cdot \omega_k)} \leq f_4 (\hat{g}_k(\epsilon)) \frac{\lambda_{ma$$

where $0 < \hat{g}_k(\epsilon) \le 1$ is the transfer function of the filter and the damping function $0 < f_4 \le 1$.

 \implies Therefore, it suffices that following inequality be **locally** hold

$$f_4\left(\hat{g}_k(\epsilon)\right) \leq \frac{1}{Re} \frac{k^2}{\lambda_{max}\left(\mathcal{S}\right)} \longrightarrow \epsilon$$

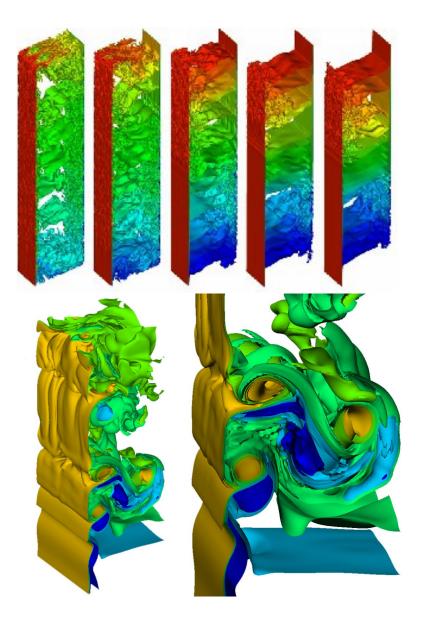
to guarantee that the **production of smaller scales of motion be stopped at the smallest scale** set by the mesh.





Previous experience with the parameter-free C_4 -regularization

Turbulence flow in a differentially heated cavity



Some details about **DNS simulations**:

- Mesh size: $128 \times 682 \times 1278$
- Computing Time: pprox 3 months 256 CPUs
- 4th-order symmetry-preserving discretization
- $A_z = 4$

Complexity of the flow:

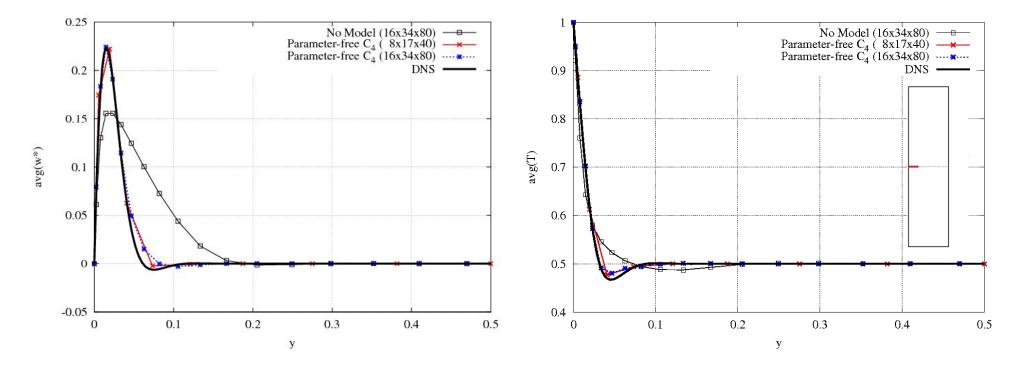
- Boundary layers
- Stratified cavity core
- Internal waves
- Recirculation areas





Previous experience with the parameter-free C_4 -regularization

Turbulence flow in a differentially heated cavity at $Ra = 10^{10}$



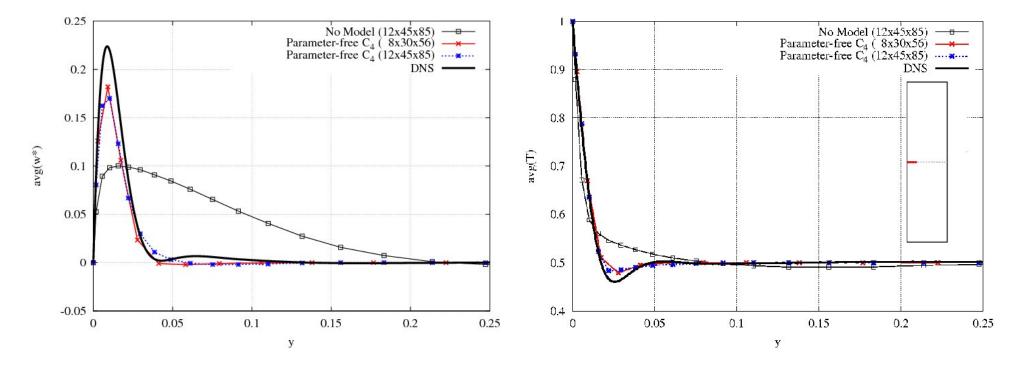
Averaged vertical velocity and temperature profiles at the horizontal mid-height plane.





Previous experience with the parameter-free C_4 -regularization

Turbulence flow in a differentially heated cavity at $Ra = 10^{11}$

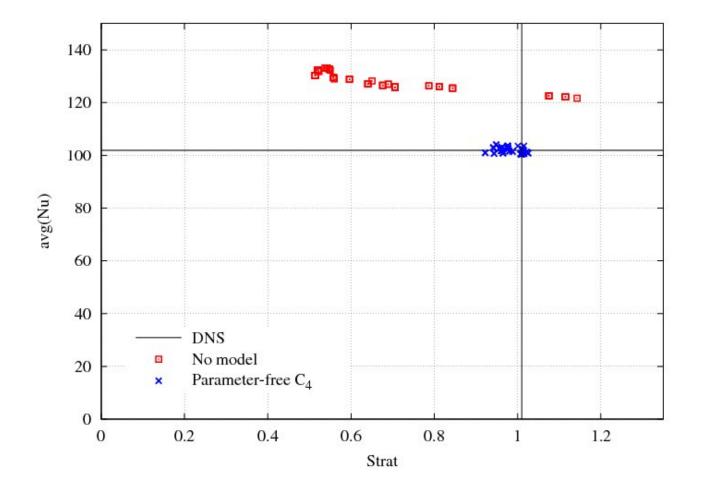


Averaged vertical velocity and temperature profiles at the horizontal mid-height plane.





A challenging test: mesh independence analysis at very coarse grids



The overall Nusselt number and the centerline stratification for 50 randomly generated coarse grids for the DHC problem at $Ra = 10^{10}$. $8 \le N_x \le 16$, $17 \le N_y \le 34$, and $40 \le N_z \le 80$.





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Results for the flow around the wall-mounted cube at Re_h = 7235
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- Parameter-free C_4 -regularization model is tested.
- Two coarse meshes are considered

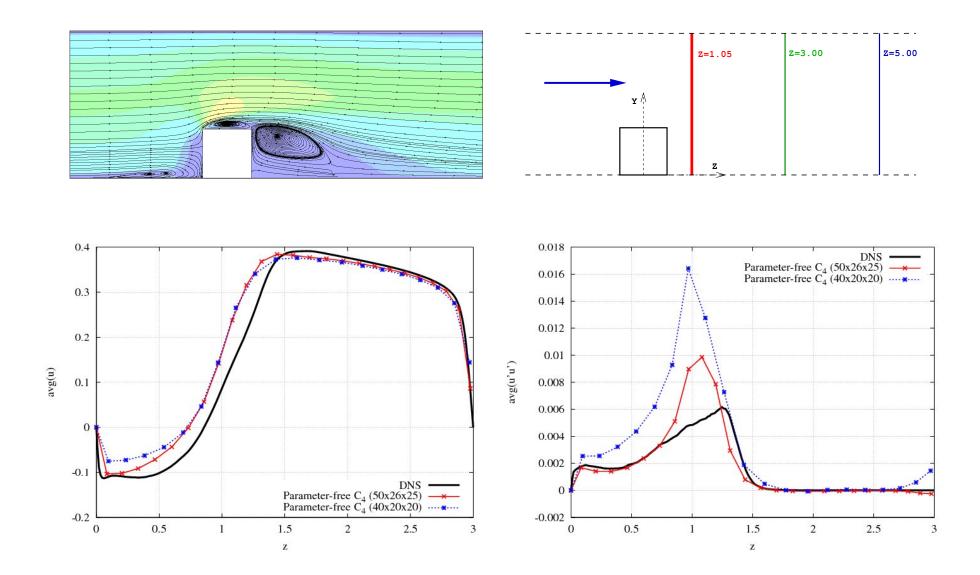
	DNS	MeshA	MeshB
Nx	400	40	50
Ny	196	20	26
$egin{array}{c} Ny\ Nz \end{array}$	200	20	25

• Coarse meshes MeshA and MeshB keep the same grid points distribution of the DNS but with **much less spatial resolutions**.





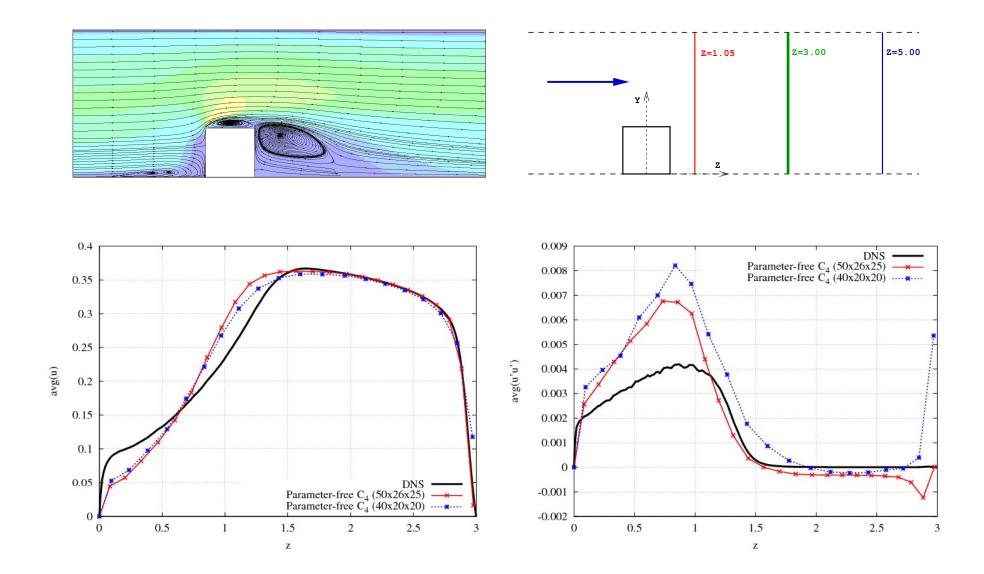
Results for the flow around the wall-mounted cube at $Re_h = 7235$







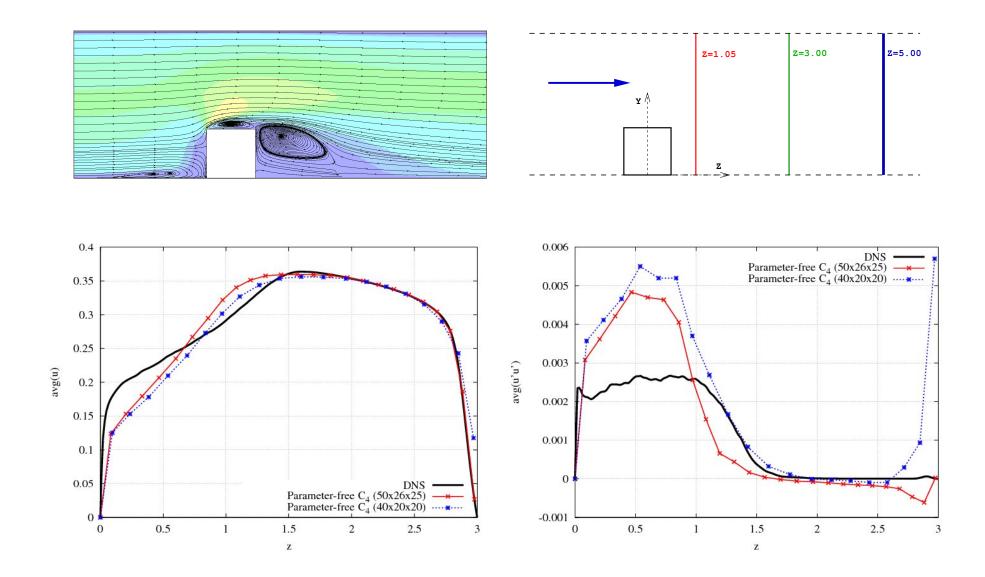
Results for the flow around the wall-mounted cube at $Re_h = 7235$







Results for the flow around the wall-mounted cube at $Re_h = 7235$







Conclusions and Future Research

The results shown illustrate the potential of the conservative parameter-free \widetilde{C}_4 smoothing as a new simulation shortcut.

The main advantages with respect exiting LES models can be summarized:

- **Robustnest**. As the smoothed governing equations preserve the symmetry properties of the original Navier-Stokes equations the solution cannot blow up (in the energy-norm, in 2D also: enstrophy-norm). It seems that even for very coarse meshes reasonably results can be obtained.
- Universality. No *ad hoc* phenomenological arguments that can not be formally derived for the Navier-Stokes equations are used.
- The proposed method constitutes a **parameter-free turbulence model**.

Since now, the method has been **successfully tested** on completely different turbulent configurations such as:

- Channel flow.
- Differentially heated cavity at different *Ra*-numbers.
- A plane impinging jet.
- Flow around a wall-mounted cube.





Thank you for you attention