

Centre Tecnològic de Transferència de Calor Laboratori de Termotècnia i Energètica UNIVERSITAT POLITÈCNICA DE CATALUNYA

VERTIMAR-2007

Numerical Simulation of the Thermal and Fluiddynamic behaviour of fuel-oil in sunk ships

C.D.Pérez-Segarra, A.Oliva, F.X.Trias, O.Lehmkuhl, R.Capdevila, Heat and Mass Transfer Technological Center (CTTC) Technical University of Catalonia (UPC) <u>www.cttc.upc.edu</u>



Edifici TR4, ETSEIAT C/ Colom, 11 08222 Terrassa (Barcelona) Tel. 93 739 81 92 Fax 93 739 81 01 segarra@cttc.upc.edu http://www.cttc.upc.edu

СТТС

Presentation:

- 1. Project objectives and work performed
- 2. Numerical studies using structured multiblock meshes.
 - a. RANS modelling.
 - b. Parallelization techniques.
- *3. Advanced turbulence modelling (DNS & LES models)*
 - a. Current DNS research at MareNostrum supercomputer.
 - b. Conservative Regularization Modelling (CRM) for LES. An overview of the method.
 - c. Results for a differentially heated cavity at $Ra=10^{10}$ and Pr=0,71.
 - *d. Results for a tank cooling problem at very high Pr-number*
- 4. Conclusions and Future research

1. Project objectives and scope of this presentation



Basic objective: Development of a code for the numerical simulation of the thermal and fluiddynamic behaviour of the fuel-oil contained in sunk ships (VEM2003-20046).

Work performed:

1. Main code based on fully-implicit structured multiblock techniques. Turbulence by means of RANS two equation models. Newtonian and non-Newtonian fluids,...

2. Development of a new advanced turbulence model.





Main code based on fullyimplicit structured multiblock techniques.



Modelling of the fuel-oil cooling inside the Prestige tanker.

- Case defined by the CIEMAT.
- 2D approach with specular symmetry on the right boundary. Left, top and bottom boundaries with vanishing velocities and constant temperature (2.6 °C). Initial temperature 50 °C.
- Thermophysical properties: ρ=1012 kg/m³;
 c_ρ=1700 J/kg K; λ=0.13 W/m K; β=7.40·10-4 K⁻¹.







Mathematical formulation (RANS models). Incompressible flows or gases at low Mach numbers.

$$\frac{1}{\rho} \frac{D\rho}{Dt} + \nabla \cdot \vec{v} = 0$$

$$\rho \frac{D\vec{v}}{Dt} \approx -\nabla \overline{p} + \nabla \cdot \vec{\tau} - \nabla \cdot \left(\rho \vec{v}' \vec{v}'\right) - \rho \beta \left(\overline{T} - T_o\right) \vec{g}$$

$$\rho c_p \frac{D\overline{T}}{Dt} \approx -\nabla \cdot \vec{q} + \nabla \cdot \left(\rho \vec{v}' T'\right)$$

Turbulent shear stresses and heat fluxes:

$$\rho \overline{\vec{v'}\vec{v'}} = -2\mu_t \overline{\vec{\gamma}} + \frac{2}{3}\rho k \vec{\delta} \qquad \rho \overline{\vec{v'}T'} = -\frac{\lambda_t}{c_p} \nabla \overline{T} = -\frac{\mu_t}{\sigma_T} \nabla \overline{T}$$



Numerical method

- *Finite volume techniques with semi-implicit differentiation in space and fully implicit in time, using Cartesian staggered grid.*
- Pressure-based method (SIMPLEC). PLDS and CDS for convective and diffusive terms respectively.
- Multiblock strategy
- Mesh: $(N_1 + N_2) \times M$. Mesh concentration near the walls. Constant Δt .

Case	N_1	N_2	M	C V	Δt	Convergence criteria	
Ref 1	100	80	200	36800	47.77	2 outer iterations	
mesh	200	160	400	145600	47.77	2 outer iterations	
dt	100	80	200	36800	4.777	2 outer iterations	
n° ite	100	80	200	36800	47.77	20 outer ite. or heat imbalance $< 1 \%$	
Ref 2	110	90	210	42840	47.77	2 outer iterations	



Computational infrastructure: PC cluster and MareNostrum supercomputer

- Low cost **PC clusters** are **loosely coupled** parallel computers:
- •Good ratio CPU power / cost
- •Low bandwidth and high latency network



Supercomputers:

- •Much bigger **number of CPU** and much higher price per CPU
- •High bandwidth and low latency network



СТТС

Some results

- High computational cost. The introduction of subdomain (or multiblock) techniques (as a parallelization strategy) allows much lower computational time. Other techniques to reduce CPUtime: variable ∆t (CFL strategies); semi-explicit methods (solid-fluid); more efficient solvers; variable grid concentration);
- Heat transfer and temperature evolution (ref 2):



СТТС

Some results:

- The phenomenology at the first 20 days is mainly convective but, once the fluid is cold enough to became very viscous, the velocity decreases and the phenomenology is diffusive predominant.
- The inner tank is almost one degree hotter than the outer (5th day).





More results (ref 2)

- 1st day: relatively high velocities at the outer tank, specially at the left boundary due to natural convection. The inferior part of the system is the cooler one, with much more vortex (initial Rayleigh = 3.72.10¹³). Mean temperature around 39 °C.
- 5th day: still movement into the tank, but considerably lower. Quite uniform temperature, around 30 °C.
- From the 20th day velocities are almost null and temperature decrease has the form of an inverse exponential.



Development of a new advanced turbulence model.



Governing equations

Incompressible Navier-Stokes coupled with the energy transport equation:

$$\nabla \cdot \vec{v} = 0$$
$$\frac{\partial \vec{v}}{\partial t} + C(\vec{v}, \vec{v}) = \mathbf{Pr} \,\nabla^2 \vec{v} - \frac{1}{\rho} \nabla p + \vec{f}$$
$$\frac{\partial T}{\partial t} + C(\vec{v}, T) = \nabla^2 T$$

Where $\vec{f} = (0,0, Ra \Pr T)$ *(Boussinesq approximation) and the nonlinear convective term is given by*

$$C(\vec{a},\vec{b}) = (\vec{a} \cdot \nabla)\vec{b}$$



Current DNS research: Differentially heated cavity at Ra=10¹¹, Pr=0.71

Direct numerical simulation Natural convection turbulent flow in a differentially heated cavity of aspect ratio 4

X. Trias, A. Gorobets, M. Soria, A. Oliva

 $Ra = 10^{11}$

Mesh size: 111M nodes

4-th order numerical approximation

Parallel computations using **512** CPU on the Supercomputer **Marenostrum** of the Barcelona Supercomputing Center

Visualisation variable: Temperature





Some details about **DNS** simulation:

•*Mesh size*: 128 x 680 x 1280 (**111·10⁶ nodes**)

•*Computing time*: ~3 months using **512 CPUs** at MareNostrum supercomputer •4th-order symmetrypreserving discretization

Complexity of the flow:

Boundary layers
Stratified cavity core
Internal waves
Recirculation areas

3. Advanced turbulence modelling (DNS & LES models)



Since the computational cost of a DNS simulation is prohibitive a **dynamically less** complex mathematical formulation is sought. To do so, we consider smooth approximations (regularizations) of the nonlinear terms

$$\frac{\partial \vec{v}_{\varepsilon}}{\partial t} + \widetilde{C}(\vec{v}_{\varepsilon}, \vec{v}_{\varepsilon}) = \mathbf{Pr} \,\nabla^2 \vec{v}_{\varepsilon} - \frac{1}{\rho} \nabla p_{\varepsilon} + \vec{f}_{\varepsilon}$$
$$\frac{\partial T_{\varepsilon}}{\partial t} + \widetilde{C}(\vec{v}_{\varepsilon}, T_{\varepsilon}) = \nabla^2 T_{\varepsilon}$$

Such approximations may fall in the Large-Eddy Simulation (LES) concept,

$$\frac{\partial \vec{v}_{\varepsilon}}{\partial t} + C(\overline{\vec{v}_{\varepsilon}}, \overline{\vec{v}_{\varepsilon}}) = \mathbf{Pr} \,\nabla^2 \overline{\vec{v}_{\varepsilon}} - \frac{1}{\rho} \nabla \overline{p_{\varepsilon}} + \overline{\vec{f}_{\varepsilon}} + M_1(\overline{\vec{v}_{\varepsilon}}, \overline{\vec{v}_{\varepsilon}})$$
$$\frac{\partial \overline{T_{\varepsilon}}}{\partial t} + C(\overline{\vec{v}_{\varepsilon}}, \overline{T_{\varepsilon}}) = \nabla^2 \overline{T_{\varepsilon}} + M_2(\overline{\vec{v}_{\varepsilon}}, \overline{T_{\varepsilon}})$$

If the model terms were given by

$$M_{1}(\overline{\vec{v}_{\varepsilon}}, \overline{\vec{v}_{\varepsilon}}) = C(\overline{\vec{v}_{\varepsilon}}, \overline{\vec{v}_{\varepsilon}}) - \overline{\widetilde{C}(\overline{\vec{v}_{\varepsilon}}, \overline{\vec{v}_{\varepsilon}})}$$
$$M_{2}(\overline{\vec{v}_{\varepsilon}}, \overline{T_{\varepsilon}}) = C(\overline{\vec{v}_{\varepsilon}}, \overline{T_{\varepsilon}}) - \overline{\widetilde{C}(\overline{\vec{v}_{\varepsilon}}, \overline{T_{\varepsilon}})}$$



The main idea behind regularization methods is to **alter the convective term** to **restrain the production of small scales** of motion by means of vortex-stretching

However, since now the existing regularization models

-Leray model

-Navier-Stokes-alpha model

Do not conserve some of the inviscid invariants:

-Kinetic energy
$$\int_{\Omega} \vec{v} \cdot \vec{v} d\Omega$$

-Enstrophy (in 2D)
$$\int_{\Omega} (\nabla \times \vec{v}) \cdot (\nabla \times \vec{v}) d\Omega$$

-Helicity (in 3D)
$$\int_{\Omega} (\nabla \times \vec{v}) \cdot \vec{v} d\Omega$$

The approximate convective operator has to be skew-symmetric

$$(\widetilde{C}(\vec{a},\vec{b}),\vec{c})=-(\widetilde{C}(\vec{a},\vec{c}),\vec{b})$$



This criterion yields the following class of approximations...

$$\frac{\partial \vec{v}_{\varepsilon}}{\partial t} + C_n(\vec{v}_{\varepsilon}, \vec{v}_{\varepsilon}) = \mathbf{Pr} \,\nabla^2 \vec{v}_{\varepsilon} - \frac{1}{\rho} \nabla p_{\varepsilon}$$

In which the convective term is smoothed according to:

$$C_{2}(\vec{a},\vec{b}) = \overline{C(\vec{a},\vec{b})}$$

$$C_{4}(\vec{a},\vec{b}) = C(\vec{a},\vec{b}) + \overline{C(\vec{a},\vec{b}')} + \overline{C(\vec{a}',\vec{b})}$$

$$C_{6}(\vec{a},\vec{b}) = C(\vec{a},\vec{b}) + C(\vec{a},\vec{b}') + C(\vec{a}',\vec{b}) + \overline{C(\vec{a}',\vec{b}')}$$

Where,

$$\vec{a}' = \vec{a} - \vec{a}$$
$$C_n(\vec{a}, \vec{b}) = C(\vec{a}, \vec{b}) + O(\varepsilon^n)$$

For any symmetric linear filter



Discretization of the convective operator: a symmetry-preserving discretization

The spatially discrete incompressible Navier-Stokes equations can be expressed as

$$\mathbf{H}\frac{d\mathbf{u}_{h}}{dt} + \mathbf{C}(\mathbf{u}_{h})\mathbf{u}_{h} + \mathbf{D}\mathbf{u}_{h} - \mathbf{M}^{\mathrm{T}}\mathbf{p}_{h} = \mathbf{0}$$
$$\mathbf{M}\mathbf{u}_{h} = \mathbf{0}$$

It can be shown that the convective matrix $C(u_h)$ has to be skew-symmetric,

$$\mathbf{C}(\mathbf{u}_{\mathrm{h}}) + \mathbf{C}^{\mathrm{T}}(\mathbf{u}_{\mathrm{h}}) = \mathbf{0}$$

To preserve the continuous invariants (kinetic energy, enstrophy in 2D and helicity in 3D) in a discrete sense.



Results for a differentially heated cavity at Ra=10¹⁰ and Pr=0,71

•DNS: mesh size: 64 x 136 x 324; 4th-order sym.-preserv discret.

•Regularization model C_4 is tested.

•Two very coarse meshes are considered:

	DNS	RM1	RM2
Nx	64	8	8
Ny	136	17	13
Nz	324	40	30

•Ratio $\frac{\mathcal{E}}{h}$ (filter length to the average grid width) is kept constant in all three spatial directions.





Results for a differentially heated cavity at Ra=10¹⁰ Mean fields

8 x 13 x 30

8 x 17 x 40



Averaged vertical velocity profile at the horizontal mid-height plane for different \mathcal{E}_h ratios



Results for a differentially heated cavity at Ra=10¹⁰ Mean fields

8 x 13 x 30

8 x 17 x 40



Averaged temperature profile at the horizontal mid-height plane for different \mathcal{E}_h ratios



Results for a differentially heated cavity at Ra=10¹⁰ Convergence studies



The maximum of the averaged vertical velocity at the horizontal mid-height plane and the overall averaged Nusselt number as a function of the $\frac{\varepsilon}{h}$ ratio







8 x 17 x 20



Day 0

T 1.00

0.91 0.82 0.73 0.64

0.55

0.36

0.18

•Even for very coarse meshes reasonable results has been obtained.

•*Temperature evolution is in* **good agreement** with the initial numerical studies.

•Since the time integration is fully explicit we are strongly limited by the Δt (CFL condition)





Evolution of the averaged temperature and the temperature at the centre of the cavity.





Evolution of the overall Nusselt at the left vertical wall and the temperature stratification at the centre of the cavity.

Conclusions and future actions

- A *CFD code* for the thermal and fluid dynamic behaviour of fuel-oil in sunk ships has been developed.
- The code is based on **structured meshes**. Complex geometries are treated using a **multiblock strategy**. Turbulence is solved by means of **RANS models**.
- Results of a benchmark case of the cooling process in the sunk tanker Prestige have been presented. **High computational effort** is required.
- A *new LES method* has been applied for a tank cooling problem at very high Pr-number

The *main advantages* with respect the existing LES models can be summerized:

- **Robustnest**. As the smoothed governing equations preserve the symmetry properties of the original NS equations the solution can not blow-up (in energy-norm; in 2D also enstrophynorm). Moreover, it seems that even for very coarse meshes reasonably results can be obtained.
- Universality. No *ad hoc* phenomenological arguments that can not be formally derived from the NS equations are used.
- → <u>Future research</u> should focus on the extension of the overall symmetry-preserving for formulation for CRM-LES to unstructured general meshes....

Moreover, a semi-implicit formulation must be implemented in order to avoid the severe Δt restrictions.

4. Final remarks



Turbulence modelling: RANS CRM-LES DNS

Numerical Methods: structured multiblock parallelization techniques

Numerical simulation of fuel-oil in sunk ships

More basic studies on Turbulence modelling DNS & CRM-LES (CTTC)

New CFD code development

Numerical methods (unstructured meshes, stronger couplings, parallelization,...) Turbulence modelling (RANS, hybrid RANS-LES, CRM-LES,...)

(TF)